

**2009 - EC 135 P2+**

**TTAF 690**



The aircraft is a P2+ with Just under 700 hrs, 7 seats, quality leather interior & exterior to VIP standard.

It also has after market TV/DVD (two screens) and a fridge that takes place of 7<sup>th</sup> seat.

***General optional equipment***

Air conditioning system for tropical environment

Paint as per Eurocopter paint

First aid kit

Multifunction handle on the main gear box cowling (LH and RH)

Tinted sun shades for cockpit windshield roof section

Tinted window for cockpit doors (light grey)

Tinted window for passenger cabin (dark grey) incl. sliding window for sliding doors

Rotor brake system

Covers for co-pilot flight controls

Engine compressor washing device

***Interior cabin lay-out***

7 Headsets Bose noise cancelling series X

Enhanced sound proofing

Separation curtain for cockpit / cabin

Sliding door fastener, intermediate and max. position, LH

Sliding door fastener, intermediate and max. position, RH

Three forward passengers seats, facing backward

Two rear passengers seats facing forward

VIP interior (leather seat. Corporate interior treatment)

***Avionics Solution 11 Single pilot IFR***

*(PINA0 code 31100)*

MEGHAS - Flight Control Display System (FCDS) single (2xSMD45)

2x MEGHAS Flight Control Display Module (FCDM - THALES)

2x MEGHAS SMD45 - pilot (PFD, ND - THALES)

2x ICP (Instrument Control Panel - THALES)

1x RCU (Reconfiguration Control Unit - THALES)

GPS / NAV / COM GNS 430, pilot (GARMIN) interfaced with FCDS

GPS / NAV / COM GNS 430, co-pilot (GARMIN) interfaced with FCDS (GPS stand-alone)

Audio/Com. Control system (pilot) AS 3100-12 (BECKER) + Intercom Select Panel (ICS selector)

Audio/Com. Control system (co-pilot) AS 3100-12 (BECKER)

Intercom amplifier IC 3100-4 (BECKER)

Marker beacon receiver/lights KR 21 (HONEYWELL)

Distance measuring equipment DMS 44 A (CHELTON/WULFSBERG)

Transponder (Mode S) GTX 330 (GARMIN)

2" Standby horizon AI 804 DC (GOODRICH), incl. emergency battery

Avionics/Radio master switches (Special ECD)

*(Minimum required optional equipment according to Sol 11)*

Slant panel

Centre console

Avionics compartment

Co-pilot flight controls

7" co-pilot instrument panel with glare shield

Map case in co-pilot door

Map case on instrument panel glare shield

Bleed air heating system

Battery, type "Saft", ULM, 40 Ah, 24 V instead of standard battery

AC system

Starter/generators (2 x 200 A, 28 V DC), inst. of std one

Reinforced rear gear cross tube (standard landing gear only)

Landing & search light (450 W)

Radar Altimeter KRA 405B (HONEYWELL)

MEGHAS sensor kit

Co-pilot static system

Digital Automatic Flight Control System – DAFCS

The engines are both the original units and have the same TTSN as the airframe. They have a 3500 hr TBO.

The BEW is 1861 KG, Longitudinal Arm 4541 mm Index 8447859 kg/mm, Lateral Arm +2 mm Index 3710 kg/mm, The empty weight configuration was with 2 crew, 5 pax seats and as is usual includes full oils and unusable fuel.

MAUW for this aircraft is 2910 kg. The SB for the upgrade to 2980 kg has not been carried out.







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